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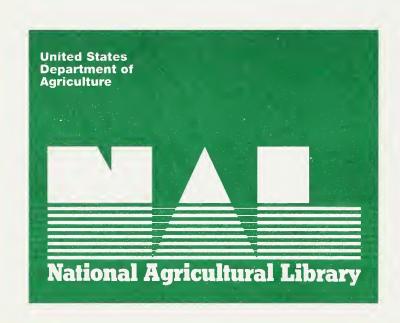
Forest Service

Technology & Development Program

5700-Aviation October 1990 9057 1202

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SELECTED INTERNAL HELICOPTER HARDPOINTS



SELECTED INTERNAL HELICOPTER HARDPOINTS

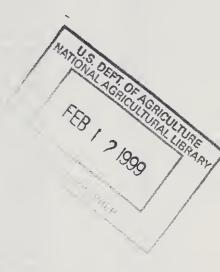
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TEO1P12
Technical Services, Aviation





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CONTENTS

		Page No.
INITOOD	NICTION	4
	UCTIONAL BACKGROUND	
	TABLES	
	5A, 212 AND 214B ATTACH POINTS AND MAXIMUM STATIC LOADS	
	DINT STRENGTHS OF ALL BELL MODELS	
	2 HARDWARE	
	6	
	NELL DOUGLAS MD500D AND MD500E	
MCDON	VELE DOGGEAG MIDSOOD AND MIDSOOL	201
	ILLUSTRATIONS	
Figure No.		Page No.
1	Load diagrams	1
2	Bell 205A floor plan	
3	Bell 212 floor plan	
4	Bell 205A bulkhead stations 129 and 166	4
5	Bell 212 bulkhead stations 129 and 166	4
6	Bell 214B bulkhead stations 129 and 166	
7	Bell 205A B.L.14 righthand wall fittings	6
8	Bell 212 B.L.14 righthand wall fittings	6
9	Bell 214B B.L.14 righthand wall fittings	
10	Bell 205A B.L.14 lefthand wall fittings	
11	Bell 212 B.L.14 lefthand wall fittings	
12	Bell 214B B.L.14 lefthand wall fititngs	
13	Bell 205A righthand roof fittings	
14	Bell 212 righthand roof fittings	
15	Bell 214B righthand roof fittings	
16	Bell 205A lefthand roof fittings	
17	Bell 212 lefthand roof fittings	
18	Bell 214B lefthand roof fittings	12
19	Bell 212 lefthand floor installation	
20	Bell 212 lefthand floor hardware list	
21	Bell 212 lefthand floor hardware list, continued	
22	Bell 212 righthand floor installation	
23	Bell 212 righthand floor hardware list	
24	Bell 212 righthand floor hardware list, continued	
25	Bell 212 floor installation	
26 27	Bell 212 floor hardware list, continued	
28	Bell 206 seat belt attach fittings	
29	Bell 206B passenger seat panel installation	
30		
31	Bell 206L pasenger seat back bulkhead installation Bell 206L baggage compartment floor installation	
32	Bell 200B bulkhead installation at station 130	
33	Bell 200L cargo hook installation	
34	Bell 200L cargo hook auxilliary equipment kit	
35	500D internal cargo loading	
36	500D memar cargo loading	
37	500D equipment installation	
38	500D (quipment installation	26
39	500E canted frame installation at station 124	27
40	500E lower section frame installation	28
. •		

TABLES

Table No.		Page No
1	Bell 205A and 212 floor fittings locations and maximum static loads	3
2	Bell 205A, 212 and 214B bulkhead stations 129 and 166 fittings locations and maximum static loads	5
3	Bell 205A, 212 and 214B B.L.14 wall fittings locations and maximum static loads	9
4	Bell 205A, 212 and 214B roof fittings locations and maximum static loads	12
5	Tensile and shear strengths of standard steel AN bolts	13



INTRODUCTION

Location and basic load bearing capacity of specified internal cargo points of Bell 205A, 212 and 214B helicopters (Bell Helicopter Textron, Ft. Worth, TX) are presented in this report. Limited information regarding location of fittings and cargo loading restrictions for McDonnell Douglas MD500C, 500D, and 500E (McDonnell DouglasHelicopter Co., Mesa, AZ), and Bell 206 and 412 helicopters is also given. Standard tiedown hardware and locations are provided, if information available. The information provided in this report is intended to assist in the safe use of internal attach points of the helicopters discussed. If any doubt exists for a specific application or loading of these locations, the helicopter manufacturer should be consulted.

GENERAL BACKGROUND

A review of engineering statics and definitions of terminology is presented here to avoid misunderstandings about the use of the maximum load tables that are presented in this report. Each attach point has a "normal" direction perpendicular to, and a "shear" direction parallel to, the surface. The directions of normal and shear have nothing to do with aircraft axes; i.e., up, down, forward, etc. Normal and shear apply to the surface to which the loads will be applied.

Loads appearing in the tables, other than normal and shear loads, are to be applied with respect to aircraft axes as noted. These directions are vertical, longitudinal, or lateralforforces of $R_{\text{VERT}}, R_{\text{LONG}},$ and $R_{\text{LAT}},$ respectively. The normal force R_{N} is always perpendicular (normal) to the attach point surface; the shear force R_{S} is parallel to the plane of the attach point surface . Therefore, for floor and roof fittings $R_{\text{N}} = R_{\text{VERT}},$ for bulkheads $R_{\text{N}} = R_{\text{LONG}},$ and for walls $R_{\text{N}} = R_{\text{LAT}}$. Also, for floor and roof fittings, $R_{\text{S}} = R_{\text{LAT}}$ or R_{LONG} (or any other force in the horizontal plane), for bulkheads $R_{\text{S}} = R_{\text{LAT}}$ or $R_{\text{VERT}},$ and for walls $R_{\text{S}} = R_{\text{LAT}}$ or $R_{\text{VERT}},$ and for walls $R_{\text{S}} = R_{\text{LAT}}$ or $R_{\text{VERT}},$ and for walls $R_{\text{S}} = R_{\text{LAT}}$ or $R_{\text{VERT}},$

USE OF TABLES

For the Bell floor fittings described in Table 1, all loads are expressed as normal to or parallel to (shear) the attach surface, or as a resultant load. The resultant is the vector sum of the normal and shear loads. The angle the resultant makes with the normal to the surface is defined as Ø. From engineering mechanics, a load can be represented either as the resultant (R) or by its components. Hence,

$$R_N = R\cos\emptyset$$
, $R_S = R\sin\emptyset$ and $R = \sqrt{R_S^2 + R_N^2}$.

R may be positive (tensile) or negative (compressive); the same load limitations on the aircraft apply. As long as either criterion I or criterion II is satisfied, the application is within the design limits of the aircraft.

An Example: Suppose a fastener on the floor has an upward load of 1500 lb, a forward load of 750 lb, and a 500 lb load toward the RHS of the aircraft. Then.

$$R_s = \sqrt{750^2 + 500^2} = 901$$
 lb (total shear load),
and $R_N = 1500$ pounds. Furthermore,
 $R = \sqrt{1500^2 + 901^2} = 1750$ pounds.

Is location 1, table A, an acceptable alternative? Refer to figure 1 for illustrated loads R, R_s, and R_N, and angle Ø. Criterion I cannot be used, since Ø>30° (\tan Ø = R_s/R_N), therefore criterion II governs. Since 1500 lb is less than 1730 lb and 901 lb <1250 lb, this loading is acceptable.

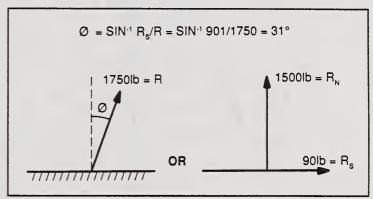


Figure 1. Load diagrams.

Example Table A.
Locations and maximum static loads.

LOCATION	CRITERION I R(lb), Ø	CRITERION II R _N & R _s MAX (lb)
Maxim	num resultant and limit	ation
1	2500, Ø≤30°	1730, 1250

For the bulkhead, wall, and roof fittings in tables 2, 3 and 4, criterion II loading combinations are dependent upon aircraft axes directions. Criterion I loadings are normal and shear combinations as in table 1, except where a general shear load does not apply. Again, if either criterion I or II is satisfied, the loading is within the aircraft design limits. Both criteria describe maximum combined loading. Load combinations from criterion I and II cannot be mixed.

Bell 205A, 212 AND 214B ATTACH POINTS AND MAXIMUM STATIC LOADS

Tables 1 through 4 of internal hardpoint data are for Bell 205A, 212 and/or 214B helicopters, as noted. These data were obtained from Bell report No. 205-099-205, Rev. C, dated August 5, 1975. Floor plan views of attach point locations for the 205A and 212 are shown in figures 2 and 3. According to Bell product data books, the Bell 212 and 412 baggage compartment and cargo floor have loading capacities of 32.3 lb/ft² and 100 lb/ft², respectively.

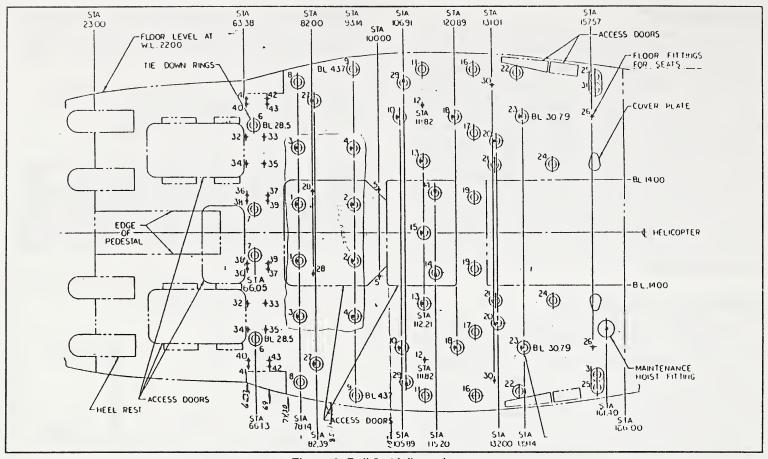


Figure 2. Bell 205A floor plan.

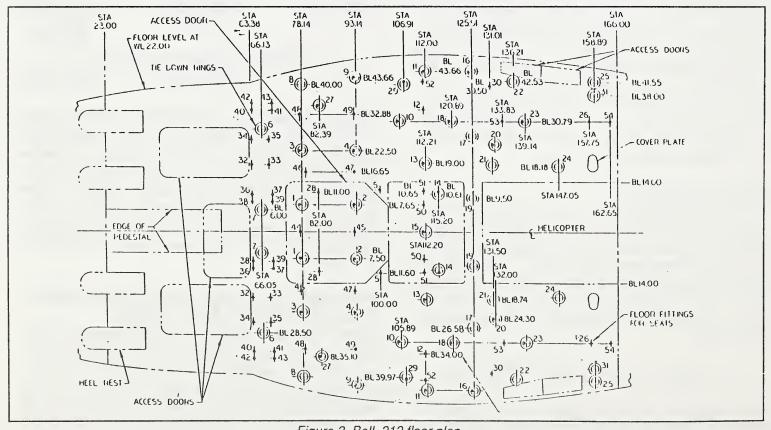


Figure 3. Bell 212 floor plan

	LOCATION	ODITEDION.	I COITEDION II
	LOCATION	CRITERION I R _N & R _s Maximun Maximum resultant and	n (lb) R(lb), Ø
	1		700 N/A
•	2		700 N/A
	3		700 N/A
	4		700 N/A
	5		N/A
	6		000 N/A
	7		000 N/A
	8		00 N/A
	9		250 2500, Ø≤30°
	10		2500, Ø≤30°
	11		2500, Ø≤30°
	12		2500, Ø≤30°
	13		2500, Ø≤30°
	14	2150 12	2500, Ø≤30°
	15	2150 12	2500, Ø≤30°
	16	2150 12	2500, Ø≤30°
	17	1250 5	500 N/A
	18	2150 12	2500, Ø≤30°
	19	1250 5	500 N/A
	20		2500, Ø≤30°
	21		500 N/A
	22		500 N/A
	23		2500, Ø≤30°
	24		N/A
	25		500 N/A
	26		2500, Ø≤30°
	27		2500, Ø≤30°
	28		2500, Ø≤30°
	29		2500, Ø≤30°
	30		2500, Ø≤30°
	31		N/A
	* 32		N/A
	* 33		N/A
	* 34		N/A
	* 35		N/A
	* 36		N/A
	* 37		N/A
	* 38		N/A
	* 39		N/A
	* 40		N/A
	* 41		N/A
			N/A
	* 42		N/A
	* 43		700 N/A
	44		700 N/A
	45		700 N/A
	46		700 N/A
	47		700 N/A
	48		700 N/A
	49		250 2500, Ø≤30°
	50		250, Ø≤30°
	51		2500, Ø≤30°
	52	— —	250 2500, Ø≤30°
	53	_	2500, Ø≤30°
	54	2150 12	2000, 2000

NOTE* For 205A: Loads to floor attach points 32 to 43 are from pilot and co-pilot inertia reel and seat belt fittings. From the Bell report for 212 helicopters, the loads still apply— but the points are provisional only.

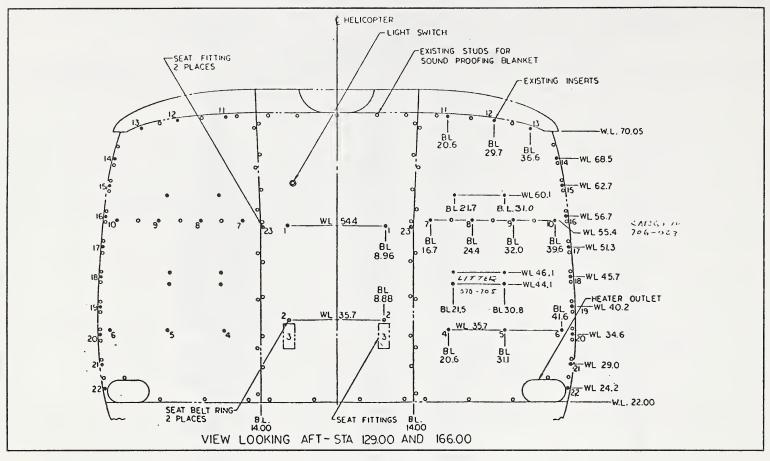


Figure 4. Bell 205A bulkhead stations 129 & 166.

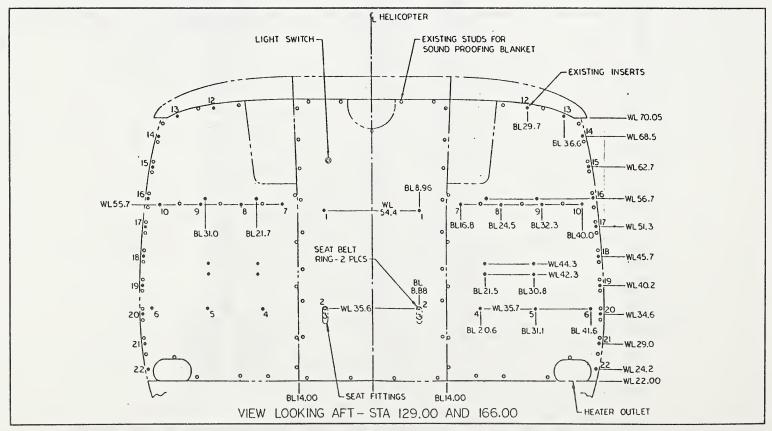


Figure 5. Bell 212 bulkhead stations 129 & 166.

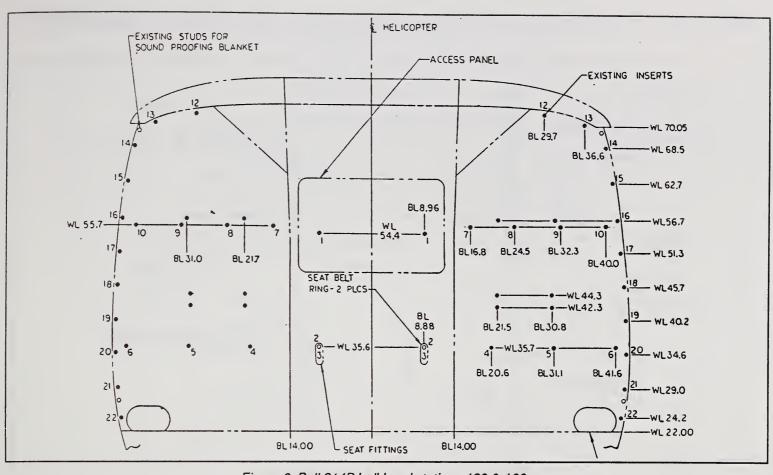


Figure 6. Bell 214B bulkhead stations 129 & 166.

Table 2.

Bell 205A, 212 and 214B bulkhead stations 129 and 166 fittings locations and maximum static loads.

LOCATION			TERION I Maximum (lb)	CRITERION II R _{vert} & R _{Lat} Maximum (lb)
* 1 * 2 * 3 4 5 6 7 8 9 10 ** 11		1250 1440 1760 1250 1250 1250 1250 1250 1250 1250 125	500 1440 0 500 500 500 500 500 500 500 1250	N/A, since R _{VERT} & R _{LAT} are R _s (see Crit. I) 1440, 1440 & R _N Maximum = 1440 1000, 0 or 0, 1000 N/A N/A N/A N/A N/A N/A N/A
13 14 15 16 17 18 19 20 21 22 23	OR,	1250 900 900 900 900 900 900 900 900 778 0	1250 300 300 300 300 300 300 300 3	N/A N/A N/A N/A N/A N/A N/A N/A N/A N/A

NOTE: * Points 1,2, and 3 are for STA 129.00; all other points are at STA 166.00 **Points 11 and 23 applicable to 205A only.

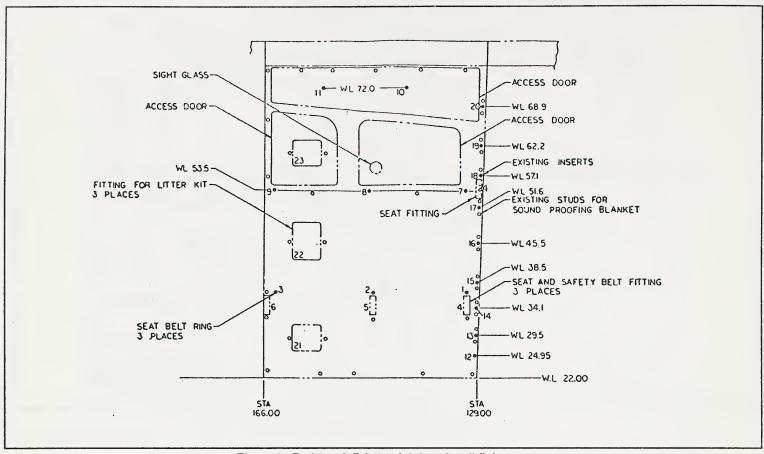


Figure 7. Bell 205A B.L.14 righthand wall fittings.

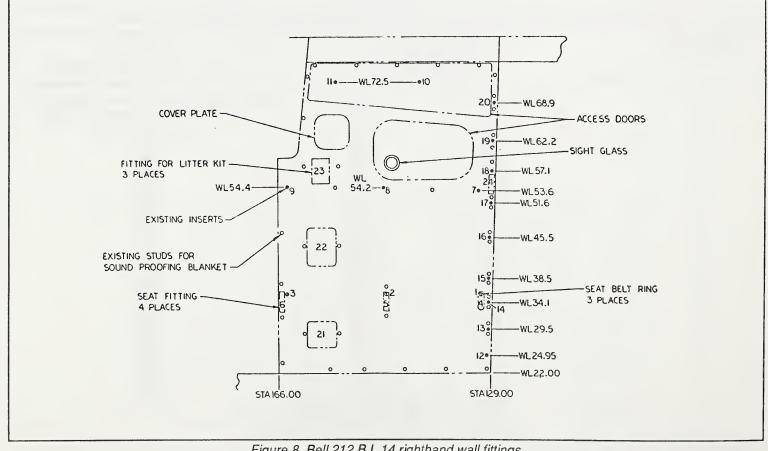


Figure 8. Bell 212 B.L.14 righthand wall fittings.

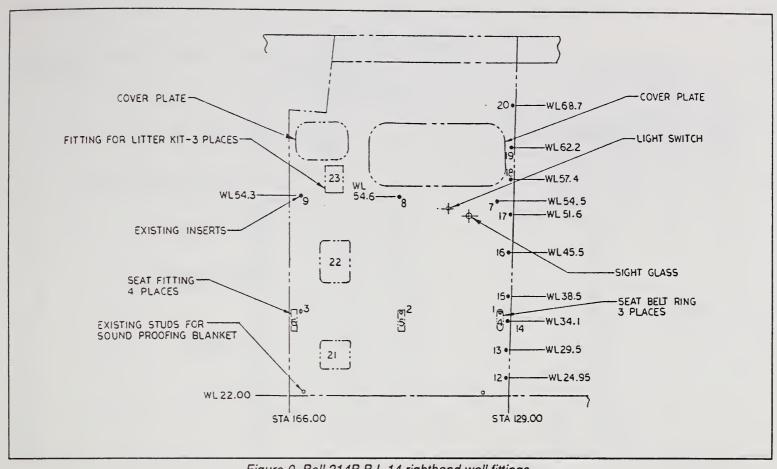


Figure 9. Bell 214B B.L.14 righthand wall fittings.

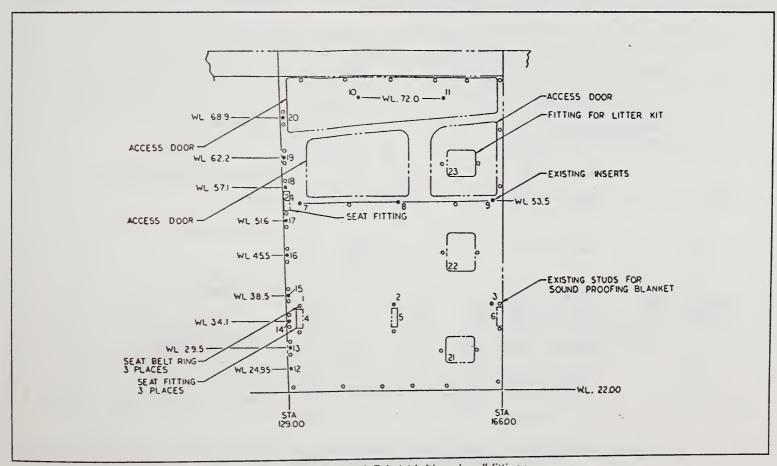


Figure 10. Bell 205A B.L.14 lefthand wall fittings.

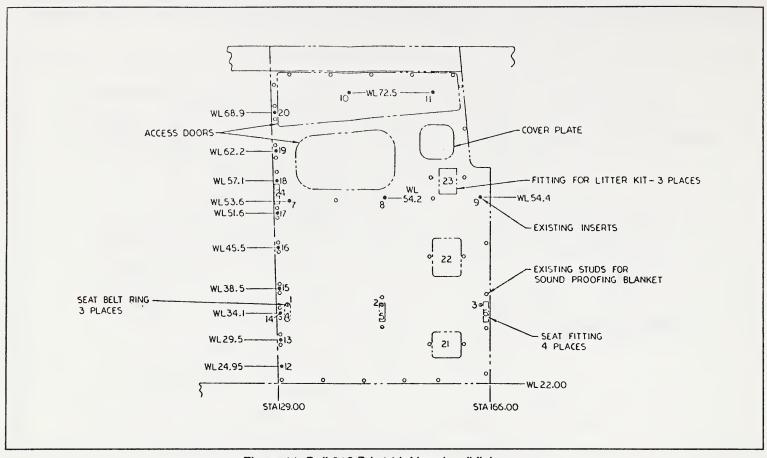


Figure 11. Bell 212 B.L.14 lefthand wall fittings.

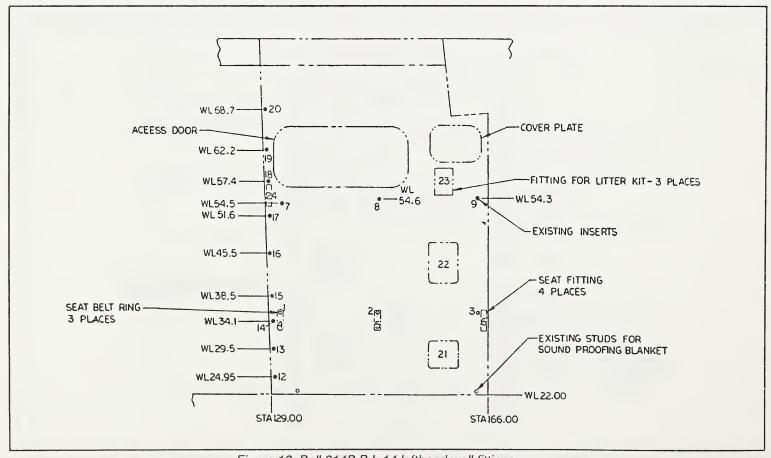


Figure 12. Bell 214B B.L.14 lefthand wall fittings.

Table 3.
Bell 205A, 212 and 214B B.L.14 wall fittings locations and maximum static loads.

LOCATION	CRITERION I R _N & R _s Maximum (lb)	CRITERION II R _{VERT} , R _{LONG} Maximum (Ib)
1 2 3 4 OR 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 * 24	1400 1400 1400 1400 1400 1400 1760 0 0 2160 0 0 0 1250 500 & R_VERT = 500 1250 500 & R_VERT = 1650 1650 1650 & R_VERT = 1650	1440 1440 & R _N Max. = 1440 1440 1440 & R _N Max. = 1440 1440 1440 & R _N Max. = 1440 1000 0 0 2160 1375 0 450 0 N/A

NOTE: Location 24 applies to attachmnets only.

Location of 205A, 212 and 214B roof fittings described in table 4 are shown in figures 13 through 18.

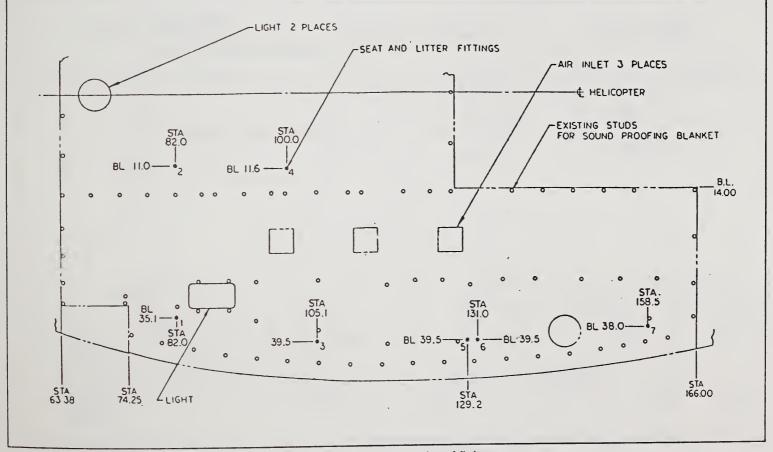


Figure 13. Bell 205A righthand roof fittings.

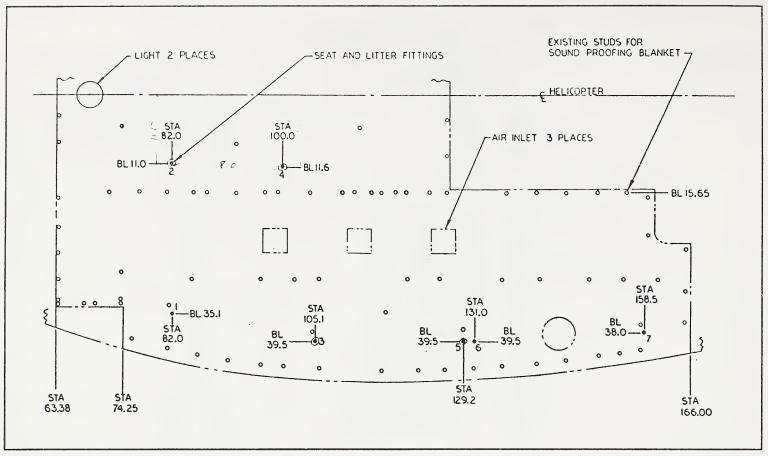


Figure 14. Bell 212 righthand roof fittings.

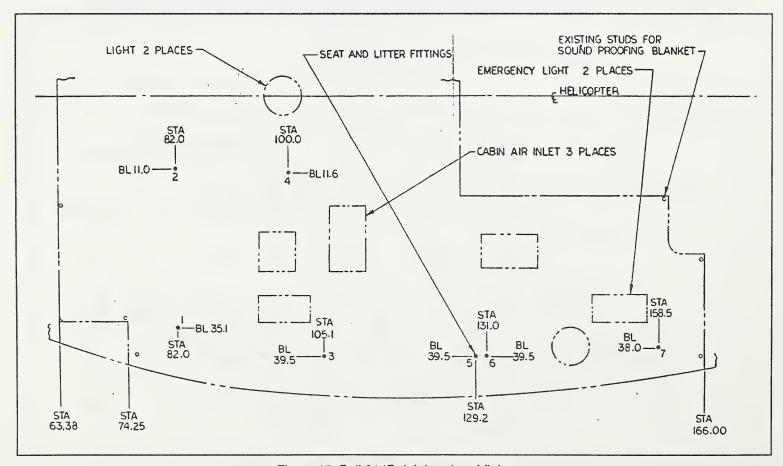


Figure 15. Bell 214B righthand roof fittings.

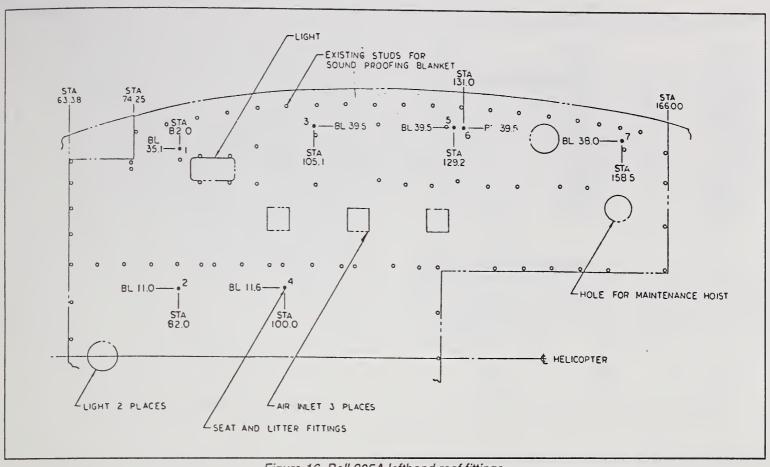


Figure 16. Bell 205A lefthand roof fittings.

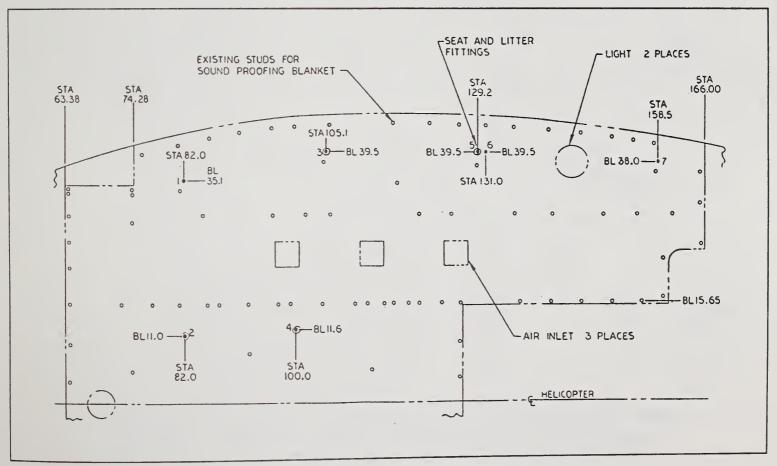


Figure 17. Bell 212 lefthand roof fittings.

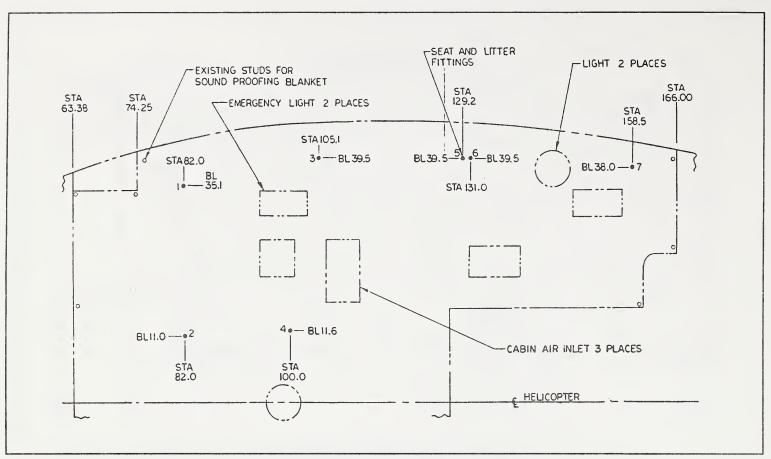


Figure 18. Bell 214B lefthand roof fittings.

Table 4
Bell 205A, 212, and 214B roof fittings locations and maximum static loads.

LOCATION			CRITERION I & R _s Maximum (lb)
1 2 3 4 5 6 7	OR OR	1500 1500 0 1250 1500 1250 (722) 0	$\begin{array}{l} {\rm R_{LONG}Max.=370} \\ {\rm R_{LAT}Max.=1130} \\ {\rm R_{LONG}Max.=370,\ R_{LAT}Max.=1130} \\ {\rm 1250} \\ ({\rm R_{LONG}Max.=344,\ R_{LAT}Max.=647,\ R_{S}=733)} \\ {\rm 1250} \\ {\rm 1250} \\ {\rm 1250} \\ {\rm 0} \end{array}$

NOTE: The bracketed values for roof points 3 and 5 of table 4 were computed for the Bell 212 rappeling operation by San Dimas Technology and Development Center (SDTDC) per letter MS-9730, dated February 28, 1979. These values were based on a 3.75 safety factor.

HARDPOINT STRENGTHS OF ALL BELL MODELS

(NOTE: Further verification from Bell engineering may be needed before the following general criteria are used.) According to the Bell Textron factory, the following statement applies to all Bell helicopter internal hardpoints for which standard AN3 (No. 10) or larger hardware could be attached. When either a shear or normal load is applied (or combination of both), the aircraft are designed so that the hardware will fail before there is any structural damage. This is not valid if any moment is applied. It should be emphasized that this is only true when considering standard AN hardware, not high strength hardware. Table 5 provides the tensile and shear strengths of standard steel AN bolts, from AN3 to AN10 inclusive. The values are from USAF T.O. 1-1A-8 table 3-1, change 26, 1 July 1969.

Table 5.
Tensile and shear strengths of standard steel AN bolts.

AN Number	Tensile (lb)	Shear (lb)
3	2210	2125
4	4080	3680
5	6500	5750
6	10,100	8280
7	13,600	11,250
8	18,500	14,700
9	23,600	18,700
10	30,100	23,000

BELL 212 HARDWARE

Bell 212 floor installations with hardware listings are shown in figures 19 through 27. These data are taken from the Bell 212 Illustrated Parts Catalog BHT-212-IPC, dated April 15, 1985.

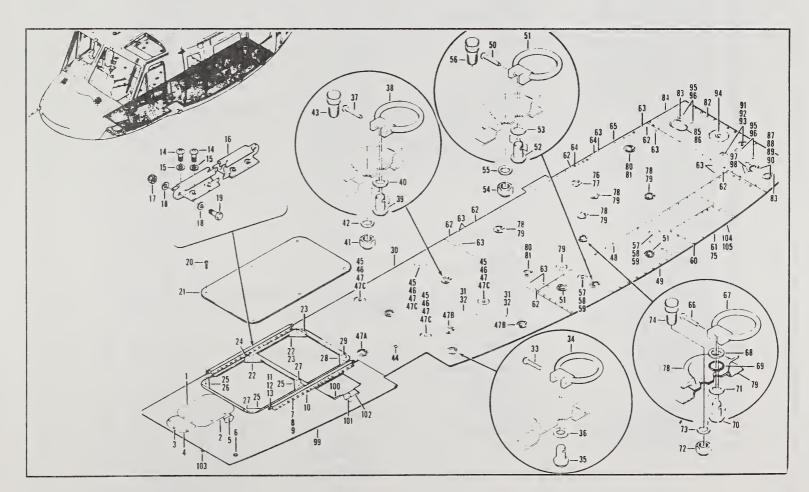


Figure 19. Bell 212 lefthand floor installation.

(1)	17	3-	141	, ¢	T.,
FIGURE &	1.6		UNIT	MODELS	"
INDEX	PART NUMBER		PER	USABLE ON	N
NUMBER			ASSY		1
					+
85 -	205-031-321-057	FLOOR INSTLILH (SEE FIG. 83A FOR NHA) (S/N	REF		,
		30504 THRU 30729)			
	205-031-321-045		REF)
	205-031-321-065	30730 THRU 30849) FLOOR INSTL,LH (SEE FIG. 83A FOR NHA) (S/N	REF		1,
	200 00 : 02 : 030	30850 AND SUB)			1
- 1	MS35207-262	SCREW	6		
- 2	204-030-321-043	DOOR	1		
- 3	MS35207-263 204-030-337-003	SCREW	3		
- 5	204-030-330-009	DOUBLER ASSY	1		
- 6	MS35489-35	GROMMET	1		
- 7	MS27039-1-11	SCREW	36		
- 8	MS27039-1-09	. SCREW	1		
- 9	AN960JD10L	. WASHER	37		
- 10	212-030-198-005	.TRACK ASSY	1		1
- 11	MS21042L3	NUT	1 2		
- 12 - 13	AN960JD10L AN3-4A	BOLT	1		
- 14	MS27039-1-11	SCREW	37		
- 15	AN960JD10L	WASHER	37		
- 16	212-030-198-001	TRACK ASSY	1		1
- 17	MS21042L3	NUT	1		
- 18	AN960JD10L	WASHER	2		
- 19	AN3-4A	BOLT	1		1
- 20	MS27039-4-08	. SCREW	6		1
- 21	212-030-250-001	.DOOR ASSY	1		
- 22	212-030-249-001	GUSSET	2		1
- 23 - 24	212-030-249-007	FILLER	1		1
- 25	212-030-249-005	GUSSET	3		1
- 26	212-030-249-011	FILLER	1		1
- 27	212-030-249-013	FILLER	2		1
- 28	212-030-249-003	. GUSSET	1		
- 29	212-030-249-015	.FILLER	1		
- 30	205-031-332-009	.FLOOR PANEL ASSY	1		
- 31	P102F10-1	PLUG (82831) (REPLACED BY 80-011P10F1-0)	2		İ
- 31 - 32	80-011P10F1-0 S102F10-08	PLUG (REPLACES P102F10-1)	2		
32	31021 10-08	80-011S10F08-0)			1
- 32	80-011S10F08-0	SLEEVE (REPLACES \$102F10-08)	2		
- 33	MS20392-1C17	PIN	2		1
- 34	110-056-1	RING ASSY	2		1
- 35	100-051-1	PIN	2		İ
- 36	AN960PD516	WASHER	2		
- 37	MS20392-1C17	PIN	4		-
- 38 - 39	110-056-1	RING ASSY	4		-
- 40	100-051-1 AN960PD516	PINWASHER	4		
- 41	MS21042L6	NUT	4		
- 42	AN960PD616	WASHER	4		
- 43	V8P69070	STUD (14821) (REPLACED BY 120-064-3HN)	4		1
- 43	120-064-3HN	STUD (REPLACES VBP69070)	4		1
- 44	204-030-036-003	PLUG	1		
- 45	MS21042L6	NUT	4		
- 46	AN960PD616	WASHER	4		
- 47	VBP69070	STUD (14821) (REPLACED BY 120-064-3HN)	4		
- 47 - 476	120-064-3HN 50-074-4	STUD (REPLACES VBP69070)	2		
	50-074-3	FITTING	4		
	50-074-1	FITTING	4		
- 48	MS27039-1-07	SCREW	104		
- 49	212-030-188-001	PANEL ASSY	1		
- 50	MS20392-1C17	PIN	3		

Figure 20. Bell 212 lefthand floor hardware list.

FIGURE	1,2	(3)	UNIT	MOD	ELS	
INDE X NUMBER	PART NUMBER	DESCRIPTION	PER	USABL		-
85 - 51	110-056-1	RING ASSY	3			+
- 52	100-051-1	PIN	3			
- 53	AN960D516	WASHER	3			
- 54	MS21042L6	NUT	3	•		
- 55	AN960D616	WASHER	3			
- 56	4159-2-11	STUD (99879)	3			
- 50	4139-2-11	(ALTERNATE PART)	3			
F.	120 064 3541					
- 5 6	120-064-3HN	STUD	3			
- 5"	MS21042L6	NUT	2			
- 5 c	AN960D616	WASHER	2			
- 5 9	4159-2-11	(ALTERNATE PART)	2			
	120 064 255		4.0			
(* >	120-064-3HN	STUD	AR			
- 60	MS27039-1-07	SCREW	17			
- t 1	212-030-182-001	COVER	1			
- + 2	AN525-10R10	.SCREW	25			ſ
- L 1	AN525 - 10R8	SCREW	75			
- E ·	AN525-10R9	SCREW	8			
- E ;	205-031-257-067	.FLOOR PANEL ASSY	1			
- 60	MS20392-1C17	PIN	8			
- 6'	110-056-1	RING	8			
- 6a	AN960-516L	WASHER	8			
- 6 9	AN622786	PACKING, PREFORMED	8			
- 70	100-051-1	PIN (00000)	8			
- 7	AN960PD516L	WASHER	8			
- 7	MS21042L6	NUT	7			
- 7	AN960PD616	WASHER	7			
- 7.	VBP69070	STUD (14821)	7			
- 7	120-064-3HN	(ALTERNATE PART)	7			
- 7	205-031-257-073	. FILLER ASSY	1			
- 7:	50-074-4	FITTING				
- 7						
- 7	205-031-2 59-0 03 50-074-3	ADAPTER	5			
- 7:	205-031-259-001	FITTING	6			
- /5 - 80		ADAPTER	- 1			
- 8C	50-074-1 205-031-259-005	FITTING	2 2			
- 82	MS27039-1-07	SCREW	38			
- 82		1	38			
- 83	MS27039-0813 205-032-147-077	SCREW	1			
- 84	MS24694S55	.FLOOR PANEL ASSY	1 4			
- 85	205-062-618-015		4			
- 85	MS20392-1C17	. CUVER ASSY	2			
- 88	110-056-1		2			
- 1		RING ASSY	2			
- 89	100-051-1	PIN (00000)	2			
- 90	AN960D516	WASHER	2		•	
- 91	MS21042L6	NUT	2			
- 92 - 93	AN960PD618 VBP69070	. WASHER	2 2			
- 93	¥8F63070	(ALTERNATE PART)	2			
- 93	120-064-3HN	STUD	2			
- 94	205-032-150-001	. BEARING SEAL	1			
	205-032-147-041	PANEL ASSY	1			
- 95	P102D8-1	PLUG (82831)	4			
		(ALTERNATE PART)				
- 95	80-011P8D1-0	PLUG	4			
- 96	\$10208-08	SLEEVE (82831)	4			
- 96	80-011S8D08-0	(ALTERNATE PART)	4			
- 97	50-074-1	FITTING	2			
- 98	205-031-350-002		4			
		FITTING	!			
- 99	205-031-200-003	.FLOOR PANEL	1			
-100	205-031-321-043	.CLIP	1			
-101	205-031-321-023	.STIFFENER	1			
	205-031-321-025	.STIFFENER	1 1			
-102		STIFFENER	1 1			
-103	205-031-321-021		1 11			
	205-031-321-021 212-030-181-001 MS21209F1-15	PAN ASSY	17			

Figure 21. Bell 212 lefthand floor hardware list, continued.

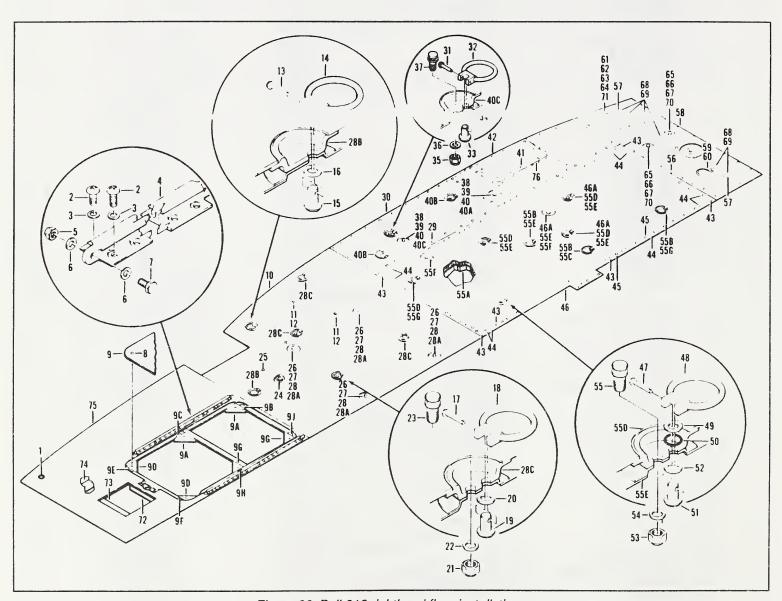


Figure 22. Bell 212 righthand floor installation.

FIGURE	(2)	اق)	UNIT	, ₅ , MODE	LS
NOE .	PART NUMBER	DESCRIPTION	PER	USABLE	
86 -	205-031-320-041	FLOOR INSTL.RH (SEE FIG. 83A FOR NHA) (S/N	REF		
	205-031-320-047	30504 THRU 30729) FLOOR INSTL,RH (SEE FIG. 83A FOR NHA) (S/N	REE		
	205-031-320-053	30730 THRU 30849) FLOOR INSTL,RH (SEE FIG. 83A FOR NHA) (S/N			1
		30850 AND SUB)			
- 1 	MS35489-35 MS27039-1-09	.GROMMET	74		
-	AN960JD10L	.WASHER	74		
- 4	212-030-198-001 MS21042L3	TRACK ASSY	2		
- £	AN9BOJD10L	WASHER	2		
- - (AN3-4A MS27039-4-08	SCREW	6		
	212-030-250-001	. DOOR ASSY	1		
	212-030-249-003	.GUSSET (USBL ON 205-031-320-41,-47)	2		
_	212-030-249-011	FILLER	1		
	212-030-249-005	GUSSET	2		
	212-030-249-013 212-030-249-011	FILLERFILLER	1		
- ,G	212-030-249-002	. GUSSET	2	^	
	212-030-249-009	.FILLER	1 !		
- 1)	212-030-249-007 205-031-332-011	FLOOR PANEL ASSY	1 1		
- 11	P102F10-1	PLUG (82831) (REPLACED BY 80-011P10F1-0)	2		1
- 1 - 1	80-011P10F1-0 S102F10-08	PLUG (REPLACES P102F10-1)	2 2		i
	31021 10 08	80-011S10F08-0)	*		
	80-011\$10F08-0	SLEEVE (REPLACES \$102F10-08)			
- 14	MS20392-1C17 110-056-1	PIN	2 2		
- 15	100-051-1	PIN	2		
- 16 - 17	AN960PD516 MS20392-1C17	WASHERPIN	2		
- 15	110-056-1	.RING ASSY	4		
- 19	100-051-1	PIN	1		
)	AN960PD516 MS21042L6	WASHER	4		
- 11	AN960PD616	WASHER	4		
- 2; - 2	VBP69070 120-064-3HN	STUD (14821) (REPLACED BY 120-064-3HN)	4		
- 2,	204-030-036-003	PLUG	1		
- 1g	204-030-036-005	PLUG	1		
- ; ;	MS21042L6 AN960PD616	NUTWASHER			
- !	VBP69070	STUD (14821) (REPLACED BY 120-064-3HN)	4		
- 1	120-064-3HN 50-074-1	STUD (REPLACES VBP69070)	4		
	50-074-4	FITTING	2		
	50-074-3	FITTING			
- 25 - 30	MS27039-1-07 212-030-188-002	.SCREW	104		
30		-47)			
- 31 - 32	MS20392-1C17	PIN	3		
- 32	100-051-1	RING ASSY	3		
- 34	AN960D516	WASHER	3		
- 35 - 36	MS21042L6 AN960D616	NUTWASHER	3		
- 37	V8P69070	STUD (14821) (REPLACED BY 120-064-3HN)	3		
- 37	120-064-3HN	. STUD (REPLACES VBP69070)	3		
- 38 - 39	MS21042L6 AN960D616	NUT	2 2		
			-		

Figure 23. Bell 212 righthand floor hardware list.

PART NUMBER	DESCRIPTION	PER ASSY		
				-
4159-2-11	STUD (99879)	2		1
VBP69070	(ALTERNATE PART)STUD (14821) (REPLACED BY 120-064-3HN)	AR		
120-064-3HN	STUD (REPLACES VBP69070)	2		-
50-074-1	FITTING	1		1
50-074-3	. FITTING	2		
		1		1
— —				
		- 1		
AN525-10R8	SCREW	74		-
AN525-10R9	. SCREW	6		
		1		1
		-		Į
AN960-516L	WASHER	8		
AN6227B6	PACKING, PREFORMED	8		
100-051-1	PIN	8		
AN960PD516L	WASHER	8		
		7		
		7		
205-031-257-074		1		
50-074-4	FITTING	1		
205-031- 259-003	ADAPTER	1		
50-074-3	FITTING	5		
_		- 1		
		38		
MS27039-0813	SCREW	8		
205-032-147-079	.FLOOR PANEL ASSY	1		
MS24694S55	SCREW	4		
		1		
		- 1		
100-051-1				
AN960D516	WASHER	2		
MS21042L6	NUT	2		
AN960PD616	WASHER			
		1 1		
		- 1		
P102D8-1	PLUG (82831)	6		
00 0440004 0	(ALTERNATE PART)		•	
	CLEEVE (02021)	1		
3 10200 - 06		Б		
80-011S8D08-0	SLEEVE	6		
50-074-1	FITTING	2		
205-031-350-002	FITTING	1		
204-031-330-001	. DOUBLER ASSY	1		
		1		
212-030-181-002	.CASTING	1		
	50-074-1 50-074-3 212-030-162-002 MS27039-1-07 212-030-182-002 AN525-10R10 AN525-10R8 AN525-10R9 205-031-257-068 205-031-225-001 MS20392-1C17 110-056-1 AN960-516L AN960-516L MS21042L6 AN960PD516L MS21042L6 AN960PD516L MS21042L6 AN960PD516L MS21042L6 AN960PD616 120-064-3HN 205-031-259-003 50-074-4 205-031-259-001 50-074-1 205-031-259-005 MS27039-1-07 MS27039-1-07 MS27039-1-07 MS27039-1-07 MS27039-1-07 MS205-032-147-079 MS24694S55 205-062-618-015 MS20392-1C17 110-056-1 100-051-1 AN960D516 MS21042L6 AN960PD616 VBP69070 120-064-3HN 205-032-147-059 P102D8-1 80-011P8D1-0 S102D8-08	50-074-1 50-074-3 5171ING 50-074-3 512-030-182-002 5CREW 5212-030-182-002 5CREW 5212-030-182-002 5CREW	SO-074-1	50-074-1 FITTING

Figure 24. Bell 212 righthand floor hardware list, continued.

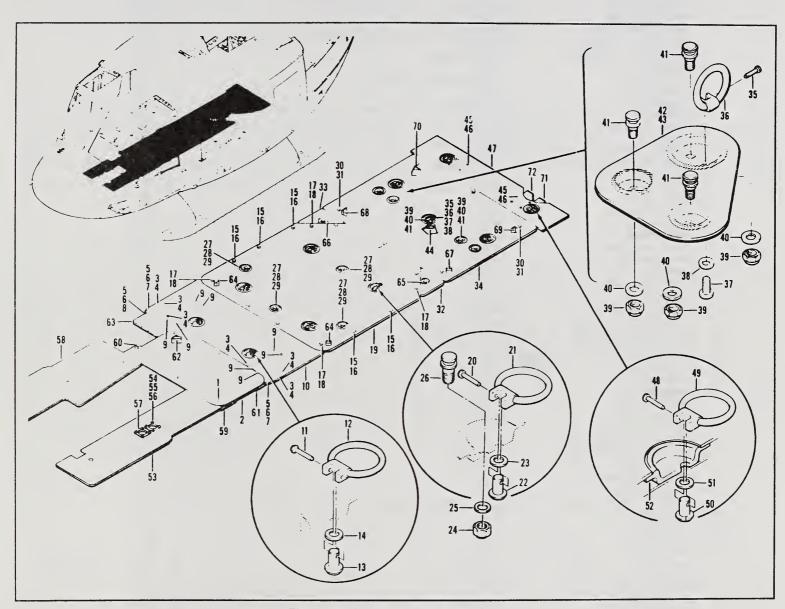


Figure 25. Bell 212 floor installlation.

1.	(2)	·3i	14.	MODELS	
FIGURE INDE	PART NUMBER	DESCRIPTION	PER	USABLE ON	ι.
NUMBE.	I AIT HOMBER	DESCRIPTION	ASSY		-
92 -	205-031-322-015	FLOOR INSTL (SEE FIG. 81 FOR NHA)	REF		
- 1	AN525-416R8 205-030-314-001	SCREW	6		
	AN525-10R9	SCREW	6		
- .	AN960JD10L	. WASHER	6		
- ·	MS2 1042L3	NUT.	3		
- c	AN960JD10L	.WASHER	3		
-	AN525-10R9 MS27039-1-09	SCREWSCREW	2		
ر -	MS27039-1-08	SCREW	8		
	205-031-258-005	.PANEL ASSY	1		
- 1	MS20392-1C17	PIN	2		
- :2	FDA1886M1	RING ASSY (98313)	2		
- :3	100-051-1 AN960D516	PIN	2		
- ,	AN4-5A	BOLT	5		
- 6	AN960JD416L	.WASHER	5		
7	AN4-4A	BOLT	9		
8	AN960JD416L	.WASHER	9		
- 9	205-031-324-007	DOOR ASSY	1		
- 10	MS20392-1C17	PIN	4		
- 2	100-051-1	.PIN	4		
- 3	AN960PD516	WASHER	4		
4	MS21042L6	NUT	4		
š	AN960PD616	WASHER	4		
ز ₋ -	120-064-3HN	STUD	4		
- 1 '	MS21042L6	NUT	4		
- 4	AN960JD416 120-064-3HN	WASHER	4		
- ĉĵ	AN4-4A	BOLT	6		
- 3	AN960JD416L	. WASHER	6		
- 3	205-030-322-067	.PANEL ASSY, LH	1		
- 3	205-030-322-068	.PANEL ASSY, RH	1		
- 3: - 3:	205-031-326-005 MS20392-1C17	DOOR ASSY	1 3		
- 3. - 3.,	110-056-1	RING ASSY	3		
- 3 /	100-051-1	PIN	3		
- 3,	AN960PD518	WASHER	3		
- 3)	MS21042L6	NUT	7		
- 4) - 41	AN960PD616 120-064-3HN	WASHER	7 7		
- 41	212-030-157-003	FITTING	1		
- 43	212-030-157-003	FITTING	1		
- 44	50-074-3	FITTING	1 1		
- 45	AN525-10R16	. SCREW	8	•	
- ·-à	AN960JD10L	.WASHER	8		
- 47 - 48	205-031-244-005 MS20392-1C17	.PANEL ASSY,FLOOR	1 2		
- 48	110-056-1	RING ASSY	2		
- 50	100-051-1	PIN	2		
- 51	AN960PD516	WASHER	2		
- £2	50-074-4	FITTING	2		
- E	205-030-322-083	.FLOOR ASSY	1		
- 5 - 5 ₂	MS21042L06 AN960JD6L	NUT	8		
- 5u	MS35206-229	. SCREW	8		
- 5	204-031-322-059	COVER	2		
- 5.,	205-030-322-087	.FLOOR ASSY	1		
- 5:	205-030-370-007	GUSSET	1		
- 60 - 6	205-030-370-011	GUSSET	1		
- 6 - 6.	205-030-370-005	GUSSET	1		
- 63	205-030-370-009	STIFFENER			
					_

Figure 26. Bell 212 floor hardware list.

FIGURE & INDEX NUMBER	PART NUMBER	DESCRIPTION	PER ASSY	MODELS 6
92 - 64 - 65 - 66 - 67 - 68 - 59 - 70 - 71 - 72	205-030-370-001 205-030-264-001 205-030-264-002 205-030-370-003 205-030-370-004 205-030-370-015 205-030-370-016 205-031-243-001 205-031-243-002	GUSSET GUSSET ASSY GUSSET ASSY GUSSET GUSSET GUSSET GUSSET GUSSET GUSSET SUPPORT FITTING SUPPORT FITTING	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	

Figure 27. Bell 212 floor hardware list, continued.

BELL 206

Per correspondence from Bell Helicopter Textron, the seat belt attach fittings are structural tie-down points. Load capability of these points was not provided by Bell Helicopter. Figure 28 shows station numbers and lateral and vertical location of these fittings for both the 206B and 206L ships. The fittings are designated Bell 206-031-104-1, -2 (LH outboard, RH outboard) and -3, -4 (LH inboard, RH inboard), respectively, for the 206B (see item 12 and 13 in figure 29), and Bell 206-032-121 fittings for the 206L series (see item 12 in figure 30).

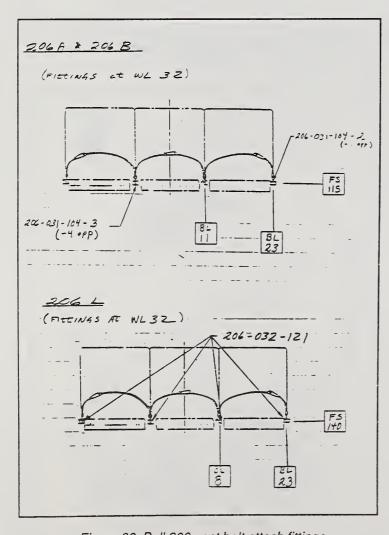


Figure 28. Bell 206 seat belt attach fittings.

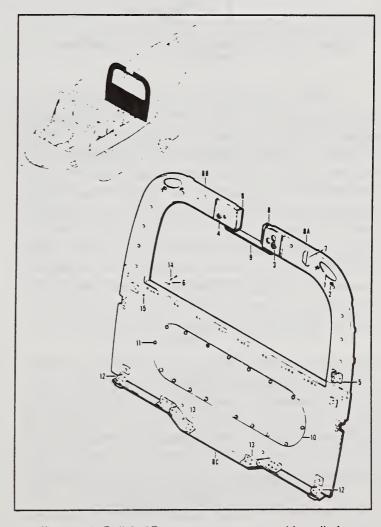


Figure 29. Bell 206B passenger seat panel installation.

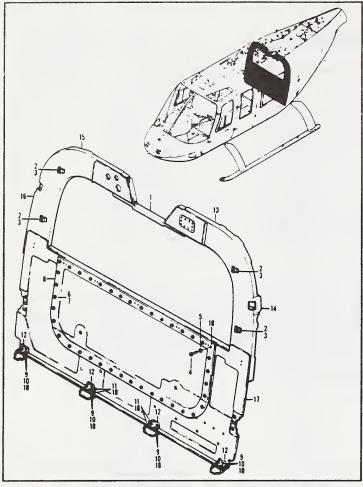


Figure 30. Bell 206L passenger seat back bulkhead installation.

From the Bell parts catalogs, there are six tiedowns on the baggage compartment floor of the 206B and nine for the 206L series. Figure 31 shows the 206L3 baggage compartment floor. The tiedowns are footman loops (P/N 90-009-4) attached to the floor with two MS24694S47 screws per loop (same hardware for all 206 series). The 206B also has four footman loops on the baggage compartment bulkhead at station 130 (see figure 32). Bell literature did not indicate tiedown strengths; however, the baggage compartment floor cargo tiedowns for the 206B military equivalent OH-58 have a strength of 2100 lb according to Army TM 55-1520-228-10.

The cargo floor load capability of the baggage compartment of Bell 206 series helicopters is 86 lb/ft²; the cabin floor and seat benches can withstand 75 lb/ft². according to Bell product data books. For 206L series helicopters fitted with the cargo hook kit, there are two fittings attached to existing holes in the rear passenger seat bulkhead with MS27039-1-08 screws (four per fitting) at B.L. 8.00 left and right, and approximately station 121.4 (figures 33 and 34). These fittings are slightly different for the 206L versus the 206L-1/-3, but use the same MS screws.

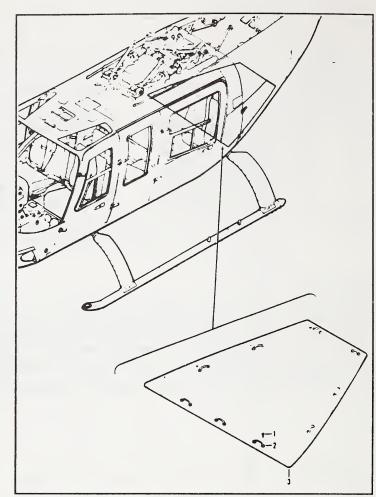


Figure 31. Bell 206L baggage compartment floor installation.

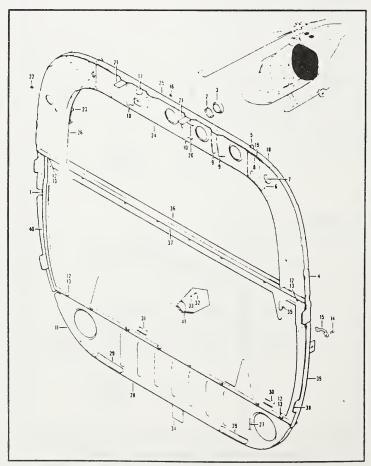


Figure 32. Bell 206B bulkhead installation at STA 130.

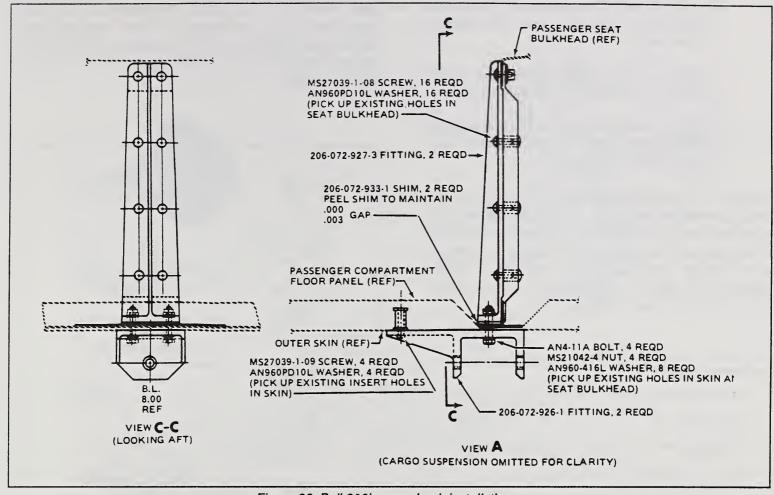


Figure 33. Bell 206L cargo hook installation.

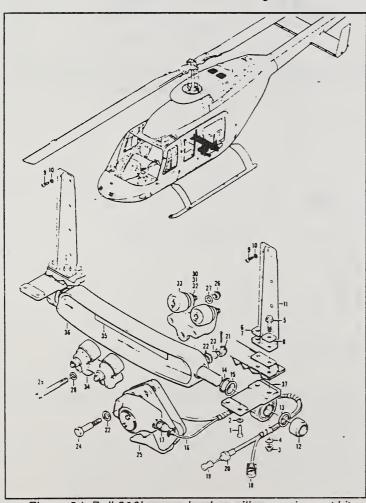


Figure 34. Bell 206L cargo hook auxiliary equipment kit.

Mc DONNELL DOUGLAS MD500D AND MD500E

Detailed internal hardpoint information for the MD500D and MD500E helicopters was not provided by McDonnell Douglas Aircraft. However, the following information was obtained from Hughes Helicopters "500D Pilot's Flight Manual" (May15,1977), and from the U.S. Army OH-6 technical manual: (*Note*: The Army OH-6 is the equivalent of the MD500C; however, the 500C, 500D and 500E are quite similar in the cargo compartment, according to Army Aviation Systems Command personnel.)

- Rope, cable, or equivalent must have a minimum loop strength of 1,800 pounds.
- Restrain the cargo from shifting by using the correct number of restraining loops in accordance with table 6-4. (figure 35).
- Position restraining loop in accordance with figure 6-7. (figure 36).
- Cargo deck capacity is 1300 lb (not to exceed 115 lb per ft²).
- View II shows typical tiedown for 500-pound cargo. (See figure 36)

- Restraining loops are to be secured as indicated and tied to the cargo to prevent slippage of the loops.
- Variations of the tiedown are allowable, providing total restraint requirements are met.
- Caution should be exercised to keep the cargo from bearing against the center slanted portion of the aft bulkhead.

The floor of the cargo compartment is a flat plate with laterally oriented hat sections. Each outboard end of the hat section has a hole drilled horizontally through the hat section with a pit pin and tie-down fitting assembly attached through each hole(figures 37 and 38). Station numbers of the cargo tie-downs (locations of hat sections) for the OH-6 are shown in figure 38.

6-8. INTERNAL CARGO LOADING

- The following should be adhered to when carrying cargo internally.
- Rope, cable, or equivalent must have a minimum loop strength of 1,800 pounds.
- Restrain the cargo from shifting, using the correct number of restraining loops in accordance with Table 6-4.
- • Position restraining loop in accordance with Figure 6-7.
- Cargo deck capacity is 1300 pounds (not to exceed 115 pounds per square foot).
- • View II shows typical tiedown for 500-pound cargo.

Table 6-4. Cargo Weight Versus Loop Requirement

Cargo (lb)	Forward Restraint	Aft Restraint	Vertical/ Lateral Restraint
Up to 100	1	1	2
101 to 300	2	1	2
301 to 400	3	2	2
401 to 600	4	2	2
601 to 800	5	3	2
801 to 1000	6	3	3
1001 to 1100	7*	4	3
1101 to 1200	8*	4	3
1201 to 1300	8*	4	3

^{*}Note the 7th and 8th loops are to use the outboard seat belt attach fitting (Station 124).

Figure 35. 500D Internal cargo loading.

Additional fittings found in the passenger/cargo area of the 500E are shown in figures 39 and 40. (*Note*: For the following fittings, -1 and -2 designate aircraft left and right respectively.) Item 16, figure 39, is designated fitting 369H3025-1, -2. Figure 40 locates the following fittings: Section A shows outboard seat support fitting 369H2509-1,-2; section B shows inboard seat support

fitting 369H2515-1, -2; section C shows seat belt fitting 369A2546-5 (1 ea.); and section D shows outboard seat belt fitting 368A2554-1, -2. [These fitting data taken from Hughes model 369E/F (MD 500E) Series IPC, Revised May 27, 1988.]

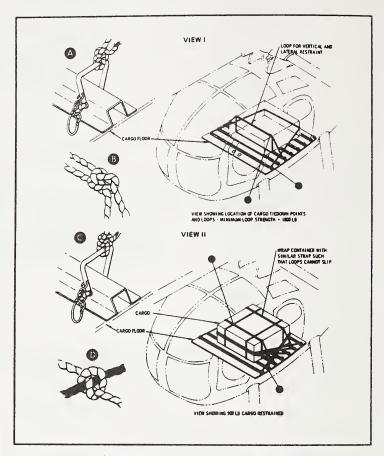


Figure 36. 500D Cargo restraint.

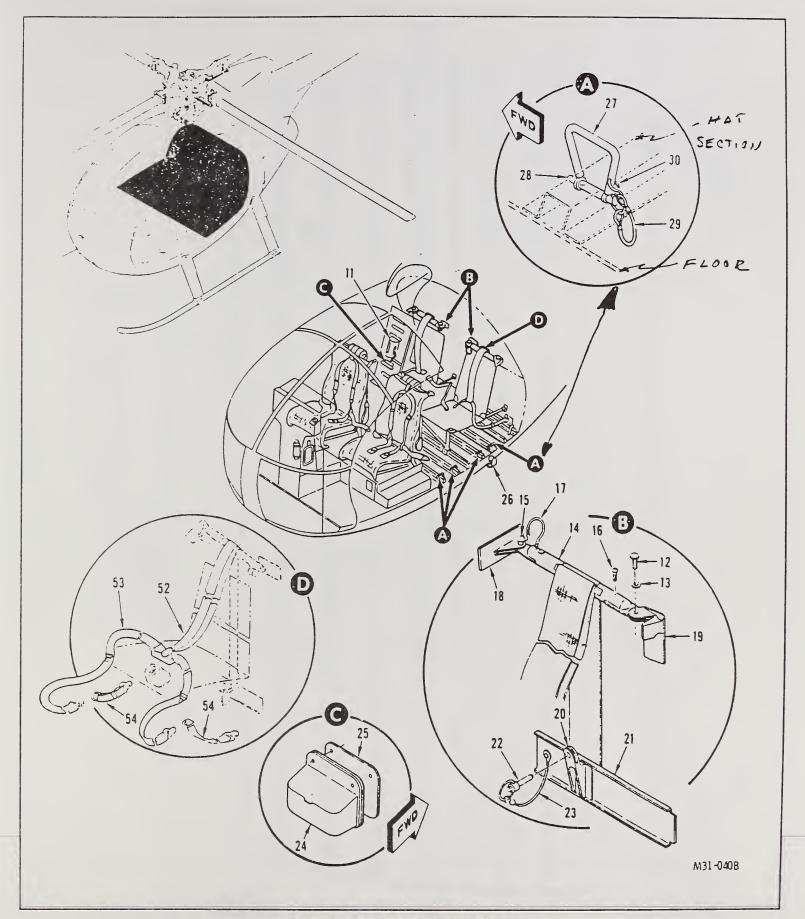


Figure 37. 500D Equipment installation.

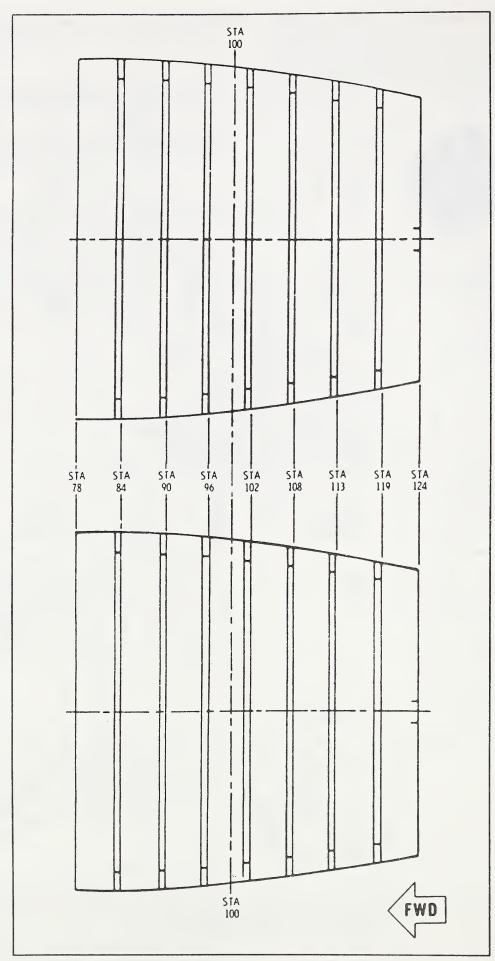


Figure 38. 500D Floor loading cg limit diagram.

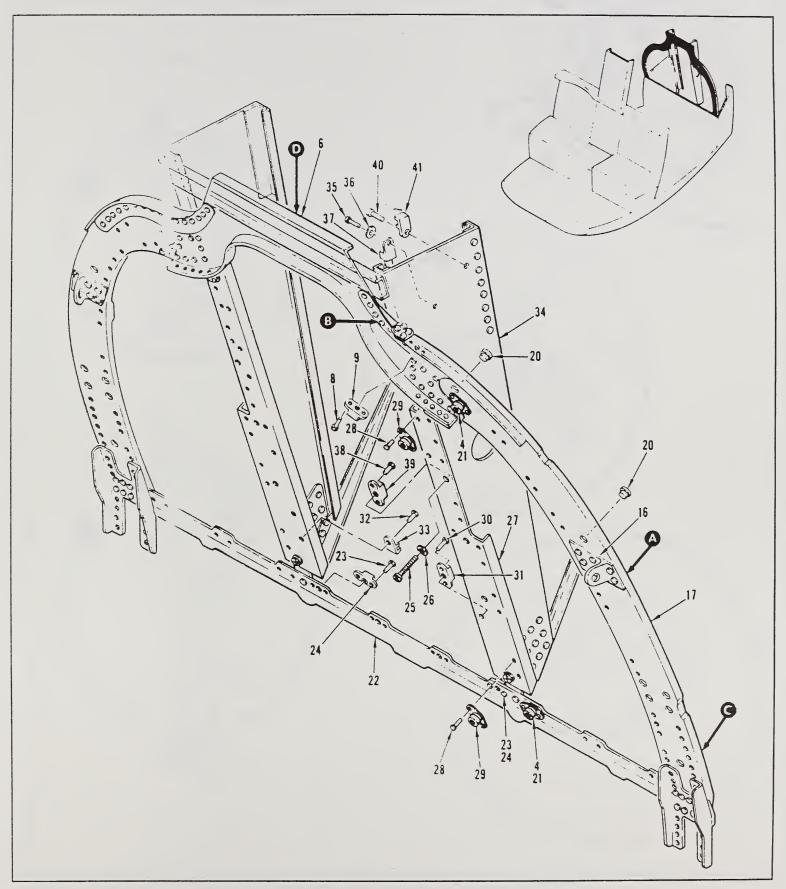


Figure 39. 500E Canted frame installation at STA 124.

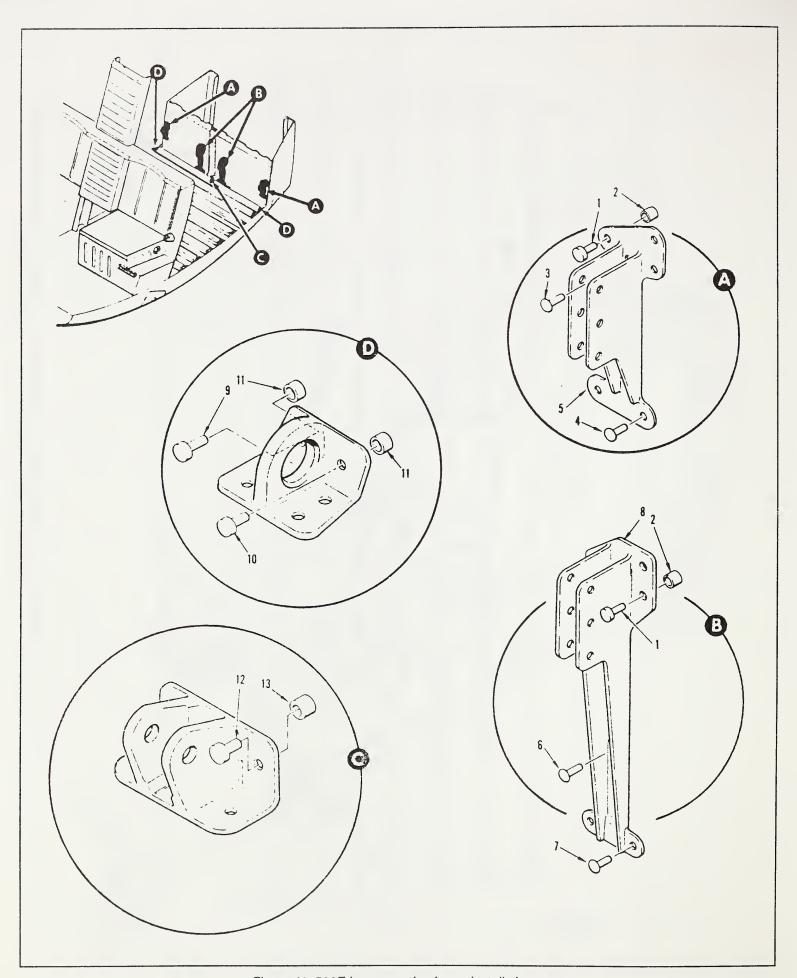


Figure 40. 500E Lower section frame installation.



